

«БОЛАШАҚҚА ҚАДАМ: ҒЫЛЫМ ЖӘНЕ ТЕХНОЛОГИЯ»
АТТЫ WCET-2012 V-ШІ БҮКІЛӘЛЕМДІК ИНЖИНИРИНГ
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1 - 2 маусым, 2012, Алматы

СБОРНИК ТЕЗИСОВ

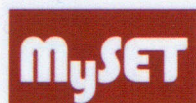
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diffuser U-bend used in a turbogenerator cooling system, a hydraulic-turbine draft tube, and an exhaust duct of a steam turbine [4]. Computations were performed on the base of the Reynolds-Averaged Navier-Stokes (RANS) equations supplemented by one or another two-equation turbulence model.

In the generic case of a U-bend, experimental study of pressure losses in an original form duct and in a CFD-optimized case was performed that confirmed a dramatic gain due to optimization. However, there was a considerable difference between the measured and predicted absolute pressure losses. This difference was even more (up to 20%) in the numerical case of the U-bend. It points that more advanced turbulence models like Large Eddy Simulation (LES)/RANS hybridization should be used at the next step of such an activity.

Regarding the problem of the draft tube shape optimization, it was stated that, apart from discretization and choice of a particular turbulence model, the flow structure and engineering quantities predicted by CFD are sensitive both to the inlet distributions of mean velocity components and to values prescribed at the inlet for transported turbulence quantities. A straight way to overcome this issue is to simulate the flow in the whole rotor/stator configuration including stay vanes, guide vanes, a runner and a draft tube. However, the high cost of time and resource consuming computations prevents introducing CFD optimization into engineering practice.

Results of CFD-based optimization of steam turbine exhaust hoods are the most optimistic despite the fact that the problem is very complicated. It turns that due to a relatively high level of turbulence and pressure gradients in an optimized shape case, neither turbulence model nor inlet turbulence level play a very significant role in the optimization output.

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ИМИТАЦИОННОЕ МОДЕЛИРОВАНИЕ РАБОТЫ ДАТЧИКА ГОРИЗОНТА

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Для ориентации относительно Земли или других небесных тел для определения местной вертикали используются датчики температуры, которые являются одними из основных датчиков, входящих в состав контрольно-измерительной аппаратуры системы ориентации космического аппарата. Датчики горизонта представляют собой инфракрасные датчики, принцип работы которых основан на измерении изменения температуры на границе космического пространства (примерно 4°K, что соответствует -269°С) и дисбаланса энергии (около 260°K для Земли), являющегося источником излучения инфракрасной энергии. Температура отмечается, когда сканирующая ось датчика пересекает линию горизонта [1], возможность определить линию горизонта небесного тела и местный вертикаль.

В работе представлена имитационная модель работы датчика горизонта с коническим сканирующим лучом в среде моделирования MathWorks MatLab/Simulink, программная реализация которой приведена на рисунке 1.

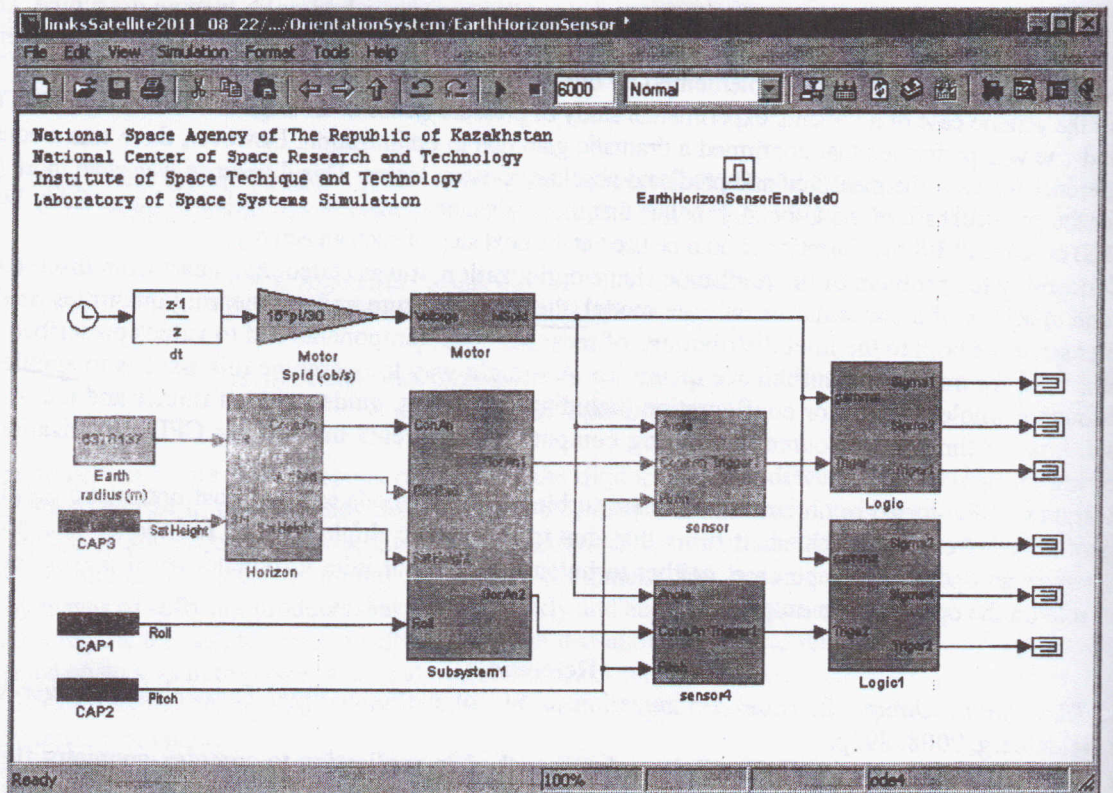


Рис. 1. Simulink-диаграмма имитационной модели «Датчик горизонта»

В работе приведены результаты моделирования датчика при определении горизонта Земли.

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SIMULATION MODELING OF THE HORIZON SENSOR

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Horizon sensor which is one of the main sensors of the measuring equipment of a spacecraft is used for determining of the local vertical near the Earth or other celestial bodies. Horizon sensors are the infrared sensors and in principle of operation is based on measuring of changes in temperature at the boundary of space (about 4 ° K, which corresponds to -269 ° C) and the disk of the planet (about 260 ° K for the Earth), which is the source of the radiation of infrared energy. The jump in temperature is noted when the scanning axis of the sensor crosses the horizon [1] and it makes possible to determine the horizon of a celestial body and the local vertical.

Simulation model of operation of horizon sensor with the cone scanning beam was developed with help of MathWorks MatLab/Simulink and its realization is showed at the figure 1.